# CanSat 2023 Post Flight Review (PFR) Outline 

\#1085
Bamantara EEPISAT

## Presentation Outline

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## Team Organization

EEPISAT


# System Overview 

## Artaka Sunu Adhi Prasetya

## Payload Design Description (1/2)

## Payload Major Components



## Information

Payload will work after released from the container. DC motor and leadscrew mechanism are used to maintain the heat shield angle. The mass is focused at the bottom of the payload to keep stability and prevent from swaying.

## Payload Design Description (2/2)

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## Container Design Description (1/2)

## Container Major Components



## Information

The descent control of the container is maintained by a parachute. The parachute has a spill hole and three side holes to improve stability and maintain nadir direction.

## Container Design Description (2/2)

## EEPISAT

Container Major Parts and Components


On/Off Switch


Bonus Camera


Compression Spring


Attachment Point



Container's Battery


Velcro Locking System

# Concept of Operations and Sequence of Events 

Fatwa Aulia Al-Haq <br> \title{
Comparison of Planned and Actual <br> \title{
Comparison of Planned and Actual CONOPS
} CONOPS
}



| CONOPS | Planned | Actual |
| :---: | :---: | :---: |
| Pre-Launch | - Arrive at the launch site <br> - GCS and antenna setup <br> - Sensor system calibration and communication with the GCS command <br> - Final CanSat check completed <br> - Activate and load CanSat into a rocket | - Arrive at the launch site $\checkmark$ <br> - GCS and antenna setup $\checkmark$ <br> - Sensor system calibration and communication with the GCS command $\sqrt{ }$ <br> - Final CanSat check completed $\checkmark$ <br> - Activate and load CanSat into a rocket $\sqrt{ }$ |
| Launch | - CanSat in a rocket launch <br> - CanSat is released from the rocket (670-725 m) <br> - Container parachute deployment with a rate of 15 m/s <br> - The video camera started to record the separation of the payload then the payload open a heat shield at 500 m with a rate of $20 \mathrm{~m} / \mathrm{s}$ or less <br> - Payload parachute deployment at 200 m with a rate of $5 \mathrm{~m} / \mathrm{s}$ <br> - Payload landed in the upright position and raised a flag 500 mm above the base of the payload. Therefore video camera stopped recording <br> - Payload shall stop transmitting data to GCS | - CanSat in a rocket launch $\checkmark$ <br> - CanSat is released from the rocket (670-725 m) $\checkmark$ <br> - Container parachute deployment with an average rate of $19.6 \mathrm{~m} / \mathrm{s}$ (in tolerance) $\checkmark$ <br> - The video camera started to record the separation of the payload then the payload open a heat shield at 500 m with an average rate of $17.5 \mathrm{~m} / \mathrm{s}$ (less than $20 \mathrm{~m} / \mathrm{s}$ ) $\checkmark$ <br> - Payload parachute deployment at 200 m with an average rate of $9.1 \mathrm{~m} / \mathrm{s}$ (out of tolerance) <br> - Payload raised flag then video camera stopped recording, but didn't in upright position <br> - Payload shall stop transmitting data to GCS $\checkmark$ |
| PostLaunch | - CanSat recovery by location from last telemetry and buzzer <br> - Inspection of CanSat damage <br> - Take the SD Card from the payload <br> - Analyze data received <br> - PFR preparation | - CanSat recovery by location from last telemetry and buzzer $\sqrt{ }$ <br> - Inspection of CanSat damage $\sqrt{ }$ <br> - Take the SD Card from the payload $\checkmark$ <br> - Analyze data received <br> - PFR preparation $\checkmark$ |

## Comparison of Planned and Actual SOE (1/2)

| CONOPS | Planned | Actual |
| :---: | :---: | :---: |
| Arrival | - Team arrival at the launch site <br> - GCS and antenna setup <br> - Check for any damages that may occur during travel | - Team arrival at the launch site <br> - GCS and antenna setup $\checkmark$ <br> - Check for any damages that may occur during travel $\sqrt{ }$ |
| Pre-Launch | - Communication inspection <br> - Mechanism inspection <br> - Assembly of the container and payload <br> - Check the CanSat dimension and weight | - Communication inspection $\checkmark$ <br> - Mechanism inspection $\checkmark$ <br> - Assembly of the container and payload $\checkmark$ <br> - Check the CanSat dimension and weight |
| Rocket Integration | - Final CanSat inspection completed before launch <br> - Turn on the CanSat, integrate it into the rocket, and ensure communication with GCS | - Final CanSat inspection completed before launch <br> - Turn on the CanSat, integrate it into the rocket, and ensure communication with GCS $\checkmark$ |

## Comparison of Planned and Actual SOE (2/2)

| CONOPS | Planned | Actual |
| :--- | :--- | :--- |

## Flight Data Analysis

Achmad Bagus Okto Faerizqi

## Payload Released at 500 Meters



[^0]
## Heat Shield Deployed

Payload Altitude


## Parachute Deployed at 200 Meters



[^1]
## Payload Altitude Plot



## Payload Temperature Sensor Plot

## EEPISAT



## Payload Battery Voltage Plot

## EEPISAT



## Information

The battery voltage had dropped when the heatshield was opening.

## Tilt Sensor Data Plot



## Payload GPS Position Plot



## Payload Camera Video



## Video properties



Video link: Click here

## Information

The camera stopped recording shortly after LANDED state achieved. That state achieved while it is still descending because the elevation of landing area is lower than the launch pad. We didn't set the camera date and time.

## Bonus Container Camera Video



Video properties


Video link: Click here

## Information

Video shown in this slide is cutted to payload release moment only. The camera started recording before the CanSat turn in the rocket and stopped after the CanSat was recovered. The payload release wasn't seen because the release moment is very fast. We didn't set the camera date and time.

## Failure Analysis

## Muhammad Tsaqif Mukhayyar

## Identification of Failures, Root Causes and Corrective Actions

| Failures | Causes | Corrective Actions |
| :--- | :--- | :--- | :--- |
| Payload is unable to upright after <br> landing | Payload hit the crop before <br> touchdown | - Change uprighting algorithm in <br> Flight Software |
| Average of payload parachute <br> descent rate doesn't meet the <br> competition requirement $(5 \mathrm{~m} / \mathrm{s})$ | -Our parachute need a lot of <br> time to gradually slow the <br> descent rate | •Correct the design for faster <br> deccelaration |
| Several GPS data loss | -GPS cannot fully lock with <br> satellites | •Add an antenna extension for <br> the GPS |

# Lessons Learned 

Fatwa Aulia Al-Haq

| What Worked | Pascions of What Worked and |
| :--- | :--- |
| Payload deployment | Payload is not in upright position |
| Payload aerobraking | Average of payload parachute descent rate doesn't <br> meet requirement |
| Payload parachute deployment | Several GPS data loss |
| Uprighting mechanism |  |
| Flag Deployment |  |
| Payload and container camera |  |
| No payload data loss |  |

## Conclusions



Bamantara EEPISAT Are Ready to be The Winner of CanSat Competition 2023

- The main objective was succeed, except the upright position
- We observed that very important to think every possibilities to prevent the mission failure
- We should consider the effect of weather and field conditions
- We learned how to work on engineering project, adapting to a teamwork environment, implementing project and time management


[^0]:    1085, 13:15:09, 904, F, DESCENT, 536.4, N, N, N, 37.2, 89.9, 8.0, 17:06:59, 683.0, 37.1952, $-80.5760,4,-4.28,98.82$, CXON
    $1085,13: 15: 10$ 905. F. .nFSCFNT. 516.1 . . . . . . . $36.8,90.1,8.0,17: 07: 00,688.0,37.1953,-80.5760,4,-20.60,112.41$, CXON
    $1085,13: 15: 11$ 906, F, HS_DEPLOYED, 499.9. P, N, N, 36.4, 90.3, 8.0, 17:07:01, 700.3, 37.1954, $-80.5760,5,-77.54,-148.69$, CXON $1085,13: 15: 12,9 \cup /, \vdash, H S$ _UEPLUYEV, $482.6, P, N, N, 35.8,90.5,8.0,17: 07: 02,711.4,37.1955,-80.5760,5,-19.31,-159.44, C X O N$ $1085,13: 15: 13,908, F, H S$ DEPLOYED, $465.6, P, N, N, 35.4,90.7,8.0,17: 07: 03,4.3,37.1956,-80.5761,5,-15.27,-144.10$, CXON

[^1]:    1085, 13:15:27, 922, F,HS_DEPLOYED, 215.2, P, N, N, 31.2, 93.5, 8.1, 17:07:17, 643.7, 37.1963, -80.5758, 5, -4.47, 11.47, CXON
     1085, 13:15:29 924, F, PC_DEPLOYED, 187.1. P, C, N, 30.8, 93.8, 8.0, 17:07:19, 625.6, 37.1964, -80.5756, 5, -3.85, -56.78, CXON $1085,13: 15: 30,925, \vdash$, PL_DEPLUYED, 1/1.1, P, C, $N, 30.6,94.0,8.0,17: 07: 20,615.0,37.1964,-80.5756,5,-15.88,-8.88$, CXON $1085,13: 15: 31,926, F$, PC_DEPLOYED, 156.5, P, C, N, 30.6, 94.1, 8.0, 17:07:21, 7.1, 37.1965, -80.5755, 5, -7.90, -2.58, CXON

