

CanSat Competition Guide 2018

Mission: Aero-Braking Atmospheric Entry Probe

**Rev 1.4
Oct 20, 2017**

Table of Contents

1. Introduction	5
1.1 Competition Description	5
2. Mission Overview	6
Concept links that inspired this mission:	7
http://marsforthemany.com/news/technology/inflatable-heat-shield/	7
http://wonderfulengineering.com/nasa-tests-foldable-heat-shields-for-future-martian-spacecrafts/	7
https://solarsystem.nasa.gov/docs/7A-8-Savino-IRENE-Preliminary-Study-Italian-Re-Entry-NaceIIE-Preliminary-Study.pdf	7
3. Requirements	8
3.1 Base Requirements	8
3.2 Selectable Bonus Objective	10
3.3 Telemetry Requirements	11
3.4 Banned Materials and components	12
4. Team Composition	13
4.1 Team Size	13
4.2 Faculty Advisor	13
4.3 Team Liaison	13
5. Deliverable Items	14
5.1 Preliminary Design Review	14
5.2 Critical Design Review	14
5.3 Post Flight Review	15
5.4 Deliverable Submissions and Scheduling	16
5.5 Slide Format Guidelines	16
5.6 Disqualification Guidelines	17
6. Flight Operations	18
6.1 Schedule	18
6.2 Flight Readiness Review	18
6.2.1 FRR Sequence of Events	18
6.3 Team Member Launch Operations Crew Assignments	19
6.4 Mission Operations Manual	19
6.5 Launch Schedule	20

6.6 Competition Operations and Sequence of Events	20
6.7 Second Flight Rules	21
6.8 Weather Delays	21
Appendix A Field Safety Rules	22
Appendix B Presentation Recommendations	23
Appendix C Payload Deployment Description	24
Appendix D Acronyms	26
Appendix E Definitions	27
Appendix F – Payload Section	29

Revisions

Date	Changes
7/5/2017	Initial guide
8/28/2017	Minor updates and date corrections. Bonus 2 updated. Bonuses given names.
9/15/2017	Reduced descent rate for heat shield aero-braking.
10/10/2017	Requirement 49 added. Section 3.3 updated with additional telemetry requirements.
10/12/2017	Expanded explanation of requirement #2
10/20/2017	Corrected Requirement 27. Updated Requirement 5,9,13. Beginning of section 2 updated.

1. Introduction

The CanSat competition is a design-build-fly competition that provides teams with an opportunity to experience the design life-cycle of an aerospace system. The CanSat competition is designed to reflect a typical aerospace program on a small scale and includes all aspects of an aerospace program from the preliminary design review to post mission review. The mission and its requirements are designed to reflect various aspects of real world missions including telemetry requirements, communications, and autonomous operations. Each team is scored throughout the competition on real-world deliverables such as schedules, design review presentations, and demonstration flights.

1.1 Competition Description

To control the size of the competition, only three teams per school are allowed to apply to the competition. It is recommended that schools hold internal design competitions to determine the three teams to apply. If more than three teams from one school apply, the first three applications received will be accepted.

The competition is in five phases.

Phase one is the application phase. Teams must submit an application and a \$100 competition fee that is non-refundable. The fee is used to offset the cost of rocket motors and other materials. Applications must be submitted by November 30, 2017. Payments will be requested in early December and must be paid by the given due date.

Phase two is the preliminary design. Teams are to develop designs, prototype, test concepts and generate a preliminary design review (PDR) slide package using the provided template. Teams will submit PDR slides in only PDF format at the designated due date. Teams that do not meet the due date or do not submit in the proper PDF format will be dropped from the competition. A schedule will be made available on when to present a subset of the slides. Teams will have a half hour to discuss a subset of the PDR slides via telecon.

After PDR, the top 40 teams will be invited to the competition.

Phase three is the critical design. Teams will finalize their design and start ordering components, manufacturing parts, test subsystems and start developing the flight unit. Teams will generate a critical design review (CDR) slide package using the provided template. Teams will submit CDR slides in only PDF format at the designated due date. Teams that do not meet the due date or do not submit in the proper PDF format will be dropped from the competition. A schedule will be made available on when to present a subset of the slides. Teams will have a half hour to discuss a subset of the CDR slides via telecon.

Phase four is the launch weekend. Friday, teams will be scheduled to present their completed CanSat for flight readiness review which must be completed in 30 minutes. The CanSat must

be ready to launch at this time. It must be completely assembled and operational. Each team will be scored during the flight readiness review. Teams can only present once at the flight readiness review at their designated time. Teams late for the review will lose points. CanSats must pass the drop test in order to fly. Multiple attempts at the drop test are allowed.

Saturday is the launch day where teams will perform final preparations and turn in CanSats by 12:00 hours local time. Launch will start at 13:00 hours local time and continue until all launches are completed. There will be no second flights unless the fault is of the launch provider and there are spare rockets and rocket motors.

Phase five is the Post Flight Review (PFR). Post Flight Review is a 15 minute presentation of the flight results and 5 minutes for questions. Awards will be presented at the end of the post Flight Reviews.

For teams to receive certificates of accomplishment and be considered for awards, they must complete all phases of the competition.

Late submissions will not be accepted for any phase and the team will be dropped from the competition. Internet issues, file issues, email issues, and any other issues will not be considered. **Submit the documents early to be safe.**

All scoring and judging results are final. Scoring is set up to be quantitative with little qualitative scoring to minimize any biases.

The competition is operated by a dedicated group of volunteers who spend their own time supporting various phases of the competition. Some volunteers spend their own funds to attend and support the competition while others are graciously supported by their employers. The competition is designed to provide teams a great educational experience and to minimize the time of the volunteers. Strict due dates, file templates, and file name formats are required to minimize the times of the volunteers who have little time to spare. Please follow all due dates and all submission requirements to help the volunteers.

2. Mission Overview

The 2018 mission simulates a space probe (CanSat) entering a planetary atmosphere. The CanSat shall consist of the probe containing the electronics and egg and a detachable heat shield. The probe shall carry a single large hen's egg. The egg must survive all portions of flight. The operation sequence shall be:

1. The probe is launched to an altitude of 670 meters to 725 meters above the launch site and then deployed from the rocket. Orientation of deployment is not controlled and is most definitely violent. Inside the rocket, the heat shield envelopes the probe completely to protect it.
2. Once the cansat is deployed from the rocket, the cansat shall open an aero-braking heat shield exposing the probe. The descent rate shall be kept at 10 to 30 meters/sec. The aero-braking probe must maintain a stable orientation with the heat shield facing the direction

of descent during descent. Tumbling is not allowed. Active control surfaces or other mechanisms can be used to maintain orientation. Stability will be verified with a tilt sensor.

3. At an altitude of 300 meters, the probe shall release the aero-braking heat shield and simultaneously deploy a parachute to reduce the descent rate of 5 meters/sec.

4. The probe shall land leaving the egg intact.

The probe shall include sensors for tracking altitude using air pressure, external temperature, battery voltage, GPS position and a tilt sensor for stability verification during descent. A compartment shall be included to hold a large hen's egg. The egg will simulate a delicate instrument.

Concept links that inspired this mission:

<http://marsforthemany.com/news/technology/inflatable-heat-shield/>

<http://wonderfulengineering.com/nasa-tests-foldable-heat-shields-for-future-martian-spacecrafts/>

<https://solarsystem.nasa.gov/docs/7A-8-Savino-IRENE-Preliminary-Study-Italian-Re-Entry-Nacelle-Preliminary-Study.pdf>

3. Requirements

3.1 Base Requirements

Requirement Number	Requirement
1	Total mass of the CanSat (probe) shall be 500 grams +/- 10 grams.
2	The aero-braking heat shield shall be used to protect the probe while in the rocket only and when deployed from the rocket. It shall envelope/shield the whole sides of the probe when in the stowed configuration in the rocket. The rear end of the probe can be open.
3	The heat shield must not have any openings.
4	The probe must maintain its heat shield orientation in the direction of descent.
5	The probe shall not tumble during any portion of descent. Tumbling is rotating end-over-end.
6	The probe with the aero-braking heat shield shall fit in a cylindrical envelope of 125 mm diameter x 310 mm length. Tolerances are to be included to facilitate container deployment from the rocket fairing.
7	The probe shall hold a large hen's egg and protect it from damage from launch until landing.
8	The probe shall accommodate a large hen's egg with a mass ranging from 54 grams to 68 grams and a diameter of up to 50mm and length up to 70mm.
9	The aero-braking heat shield shall not have any sharp edges to cause it to get stuck in the rocket payload section which is made of cardboard.
10	The aero-braking heat shield shall be a florescent color; pink or orange.
11	The rocket airframe shall not be used to restrain any deployable parts of the CanSat.
12	The rocket airframe shall not be used as part of the CanSat operations.
13	The CanSat, probe with heat shield attached shall deploy from the rocket payload section.
14	The aero-braking heat shield shall be released from the probe at 300 meters.
15	The probe shall deploy a parachute at 300 meters.
16	All descent control device attachment components (aero-braking heat shield and parachute) shall survive 30 Gs of shock.

17	All descent control devices (aero-braking heat shield and parachute) shall survive 30 Gs of shock.
18	All electronic components shall be enclosed and shielded from the environment with the exception of sensors.
19	All structures shall be built to survive 15 Gs of launch acceleration.
20	All structures shall be built to survive 30 Gs of shock.
21	All electronics shall be hard mounted using proper mounts such as standoffs, screws, or high performance adhesives.
22	All mechanisms shall be capable of maintaining their configuration or states under all forces.
23	Mechanisms shall not use pyrotechnics or chemicals.
24	Mechanisms that use heat (e.g., nichrome wire) shall not be exposed to the outside environment to reduce potential risk of setting vegetation on fire.
25	During descent, the probe shall collect air pressure, outside air temperature, GPS position and battery voltage once per second and time tag the data with mission time.
26	During descent, the probe shall transmit all telemetry. Telemetry can be transmitted continuously or in bursts.
27	Telemetry shall include mission time with one second or better resolution. Mission time shall be maintained in the event of a processor reset during the launch and mission.
28	XBEE radios shall be used for telemetry. 2.4 GHz Series 1 and 2 radios are allowed. 900 MHz XBEE Pro radios are also allowed.
29	XBEE radios shall have their NETID/PANID set to their team number.
30	XBEE radios shall not use broadcast mode.
31	Cost of the CanSat shall be under \$1000. Ground support and analysis tools are not included in the cost.
32	Each team shall develop their own ground station.
33	All telemetry shall be displayed in real time during descent.
34	All telemetry shall be displayed in engineering units (meters, meters/sec, Celsius, etc.)
35	Teams shall plot each telemetry data field in real time during flight.
36	The ground station shall include one laptop computer with a minimum of two hours of battery operation, XBEE radio and a hand held antenna.
37	The ground station must be portable so the team can be positioned at the

	ground station operation site along the flight line. AC power will not be available at the ground station operation site.
38	Both the heat shield and probe shall be labeled with team contact information including email address.
39	The flight software shall maintain a count of packets transmitted, which shall increment with each packet transmission throughout the mission. The value shall be maintained through processor resets.
40	No lasers allowed.
41	The probe must include an easily accessible power switch.
42	The probe must include a power indicator such as an LED or sound generating device.
43	The descent rate of the probe with the heat shield deployed shall be between 10 and 30 meters/second.
44	The descent rate of the probe with the heat shield released and parachute deployed shall be 5 meters/second.
45	An audio beacon is required for the probe. It may be powered after landing or operate continuously.
46	Battery source may be alkaline, Ni-Cad, Ni-MH or Lithium. Lithium polymer batteries are not allowed. Lithium cells must be manufactured with a metal package similar to 18650 cells.
47	An easily accessible battery compartment must be included allowing batteries to be installed or removed in less than a minute and not require a total disassembly of the CanSat.
48	Spring contacts shall not be used for making electrical connections to batteries. Shock forces can cause momentary disconnects.
49	A tilt sensor shall be used to verify the stability of the probe during descent with the heat shield deployed and be part of the telemetry.
50	

3.2 Selectable Bonus Objective

1. Camera: Add a color video camera to capture the release of the heat shield and the ground during the last 300 meters of descent. The camera must have a resolution of at least 640x480 and a frame rate of at least 30 frames/sec. The camera must be activated at 300 meters.
2. Wind Sensor: A radio transmitter shall be added to transmit the wind speed by changing its

frequency. The frequency change shall be 1 Hz per 0.1 meter/sec. The transmitter must be custom designed and built. It cannot be a commercial product. The frequency must be in the 433 MHz ISM band or if a team member has an amateur radio license, an amateur radio band can be used. The transmitter must be able to be set to 8 different frequencies in the 433 MHz ISM band with 25 KHz separation. The transmitter must turn off after the probe lands to minimize interference. The team can use a commercial receiver.

3.3 Telemetry Requirements

Upon powering up, the CanSat probe shall collect the required telemetry at a 1 Hz sample rate. The telemetry data shall be transmitted with ASCII comma separated fields followed by a carriage return in the following format:

<TEAM ID>,<MISSION TIME>,<PACKET COUNT>,<ALTITUDE>, <PRESSURE>,
<TEMP>,<VOLTAGE>,<GPS TIME>,<GPS LATITUDE>,<GPS LONGITUDE>,<GPS
ALTITUDE>,<GPS SATS>,<TILT X>,<TILT Y>,<TILT Z>,<SOFTWARE STATE>

1. The probe telemetry for the entire mission shall be saved on the ground station computer as a comma separated value (.csv) file that will be examined by the competition judges in Excel. Teams will provide the file to the judges after the launch operations via USB drive.

2. The telemetry data file shall be named as follows:

CANSAT2018_TLM_<TEAM_ID>_<TEAM_NAME>.csv

where the team_id is the four digit team id number and <team_name> is to be selected by each team (using underscores instead of spaces in the file name).

3. <TEAM ID> is the assigned team identification.

4. <MISSION TIME> is the time since initial power up in seconds.

5. <PACKET COUNT> is the count of transmitted packets, which is to be maintained through processor reset.

6. <ALTITUDE> is the altitude with one meter resolution.

7. <PRESSURE> is the measurement of atmospheric pressure.

8. <TEMP> is the sensed temperature in degrees C with one degree resolution.

9. <VOLTAGE> is the voltage of the CanSat power bus.

10. <GPS TIME> is the time generated by the GPS receiver.

11. <GPS LATITUDE> is the latitude generated by the GPS receiver.

12. <GPS LONGITUDE> is the longitude generated by the GPS receiver.

13. <GPS ALTITUDE> is the altitude generated by the GPS receiver.

14. <GPS SATS> is the number of GPS satellites being tracked by the GPS receiver.

15. <TILT X> Tilt sensor X axis value.
16. <TILT Y> Tilt sensor Y axis value.
17. <TILT Z> Tilt sensor Z axis value.
18. <SOFTWARE STATE> is the operating state of the software. (boot, idle, launch detect, deploy, etc.)
19. Additional data fields may be appended after the required fields as determined necessary by the team's design

3.4 Banned Materials and components

1. No foam based beads or other similar bits of foam material that can be dropped and lost on the ground. This material is dangerous to the livestock that occupy this area.
2. No lithium polymer batteries. The battery is relatively easy to damage and a fire hazard. We want to avoid setting any parts of the field on fire.

4. Team Composition

Students currently enrolled in undergraduate degree programs, or students having graduated from such programs since the start of the current competition cycle, are counted as undergraduate students.

Students currently enrolled in post-graduate degree programs (MS, PhD), or students having entered such programs since the start of the current competition cycle, are counted as graduate students.

4.1 Team Size

Each team shall consist of between 3 and 10 students (undergraduate teams) from an accredited college or university. Teams may consist entirely of undergraduate students (undergraduate teams), entirely of graduate students (grad teams), or a combination of the two (mixed teams). Graduate teams shall consist of no more than 5 students. Mixed teams shall consist of no more than 7 undergraduate students and 3 graduate students.

Teams from the same school must develop their designs independently and **not** copy from other teams. Bulk purchasing of materials is allowed such as batteries and raw materials for construction. Sharing tools and services are allowed. Designs must originate from within the team.

There shall be no more than three teams from any one school.

4.2 Faculty Advisor

Each team must have a faculty advisor. The role of the faculty advisor is to:

- Provide a point of contact for the team, both with the university and the competition.
- Aid teams with logistics such as arranging conference rooms, laboratory resources, etc.
- Providing general guidance throughout the competition.

The faculty advisor shall not:

- Make design decisions or direct recommendations.
- Participate in more than an oversight role during reviews.

4.3 Team Liaison

Each team will be assigned a competition liaison by the competition committee who acts as a liaison between the team and the competition committee. The liaison will be responsible for scheduling all competition reviews and coordinating all communications with the team. Liaisons are also responsible for tracking the team's progress throughout the competition.

Team liaisons are available to answer questions and provide general guidance. The liaison shall not provide design recommendations.

5. Deliverable Items

Teams will be evaluated based on a series of deliverable items provided at various stages of the development. The deliverable items are selected to provide representative real-world milestones for tracking the CanSat development and ensuring team success.

5.1 Preliminary Design Review

The PDR is a “multi-disciplined technical review to ensure that the system under review can proceed into detailed design, and can meet the stated performance requirements within cost (program budget), schedule (program schedule), risk, and other system constraints”. The CanSat PDR shall demonstrate:

- An understanding of the CanSat mission requirements
- Allocation and derivation of system and subsystem requirements
- Definition of the CanSat concept of operations
- Overview of preliminary design that meets specified requirements
- Results of, or identification of, necessary trades to support preliminary design. While it is ideal to have completed trades prior to the preliminary design, it is not necessary.
- Results of, or identification of, necessary prototyping or testing efforts necessary to support or finalize the preliminary design.
- Preliminary budget
- Detailed development schedule

Preliminary design reviews shall be conducted via teleconference coordinated by the team lead(s) and mentors. The PDR presentations shall be less than 30 minutes in duration including time for questions. Presentation reviewers shall be permitted to ask questions during the presentation (i.e., questions are not held until the end of the presentation).

The PDR shall follow the presentation template posted on the CanSat Competition website.

5.2 Critical Design Review

The CDR is “a multi-disciplined technical review to ensure that the system under review can proceed into system fabrication, demonstration, and test; and can meet the stated performance requirements within cost (program budget), schedule (program schedule), risk, and other system constraints”. The CDR shall demonstrate:

- All PDR level requirement TBDs and TBRs shall be resolved
- Refinement of the CanSat CONOP
- Results of detailed design and analysis for each subsystem
- Verification that detailed design meets system and subsystem level requirements

- Identification of subsystem and system level tests necessary for requirements verification
- Results of requirements verification tests completed to date
- Overview of mission operations
- Preliminary launch day sequence of events
- Revised budget
- Updated development schedule

Critical design reviews shall be conducted via teleconference coordinated by the team lead(s) and mentors. The CDR presentations shall be less than 30 minutes in duration including time for questions. Presentation reviewers shall be permitted to ask questions during the presentation (i.e., questions are not held until the end of the presentation).

The CDR shall follow the presentation template specified in the "CanSat 2018 CDR Outline" document available on the CanSat Competition website. Extra material in the form of backup slides is permitted.

Each section of the CDR shall be scored in accordance with the values listed in the outline. The CDR shall contribute to the total evaluation of the CanSat design according to the values listed the section Evaluation and Scoring.

5.3 Post Flight Review

The PFR provides an assessment of flight operations and results of the demonstration flight. The PFR provides an assessment of successful and unsuccessful flight operations. The PFR shall provide:

- Overview of mission objectives and CanSat design
- Comparison of planned and actual CONOPS and SOE
- Raw and processed data from flight operations
- Failure analysis and assessment (for unsuccessful mission objectives)

Post Flight Reviews shall be conducted the day following the demonstration flight activities, unless flight operations are canceled due to weather. Presentations shall be limited to 20 minutes, including questions.

Each section of the PFR shall be scored in accordance with the values listed in the outline. The PFR shall contribute to the total evaluation of the CanSat design according to the values listed in the section Evaluation and Scoring.

Post Flight Review presentations shall be submitted by 8:45 AM to the judges. There will be two presentation rooms. Teams will be preassigned to the presentation room. Late submissions will lose points. Each team will be given a thumb drive. Teams are to install their PFR slides on the thumb drive and deliver the thumb drive to the designated location between 8 AM and 8:45 AM Sunday. Teams delivering after 8:45 AM will lose points. This is done to

make sure all teams have the same amount of time to prepare for PFR.

5.4 Deliverable Submissions and Scheduling

All deliverable items shall be submitted to the team mentor by the dates listed in Table 1. All deliverable items shall be submitted in PDF format using the naming convention listed in Table 1 where # corresponds to the assigned team number for each team and v# is a unique revision number for the review package that can be used to track revisions. For example, a submission for Team number 1021 of version 2 of the PDR package would be named **Cansat2017_1021_PDR_v02.pdf**. Note that adherence to the filename and format specification is scored during the competition.

Presentations will be scheduled after submission of the document. A calendar of available time slots will be sent to all teams. Each team is to send to their mentor a list of three time slots. The mentor will schedule the presentation time and notify the team.

Updated presentations will not be accepted after the deadline. It is understood and expected that changes will occur between document submission and the presentation time. The scoring is based on the quality of the presentation and understanding of the competition requirements. There will be no point loss due to changes in the design between document submission time and the presentation time.

Table 1: Deliverable item due dates

Deliverable

Material Due	Required Filename Format	Due Date
PDR	Cansat2018_XXXX_PDR_vYY.pdf	02/01/18
CDR	Cansat2018_XXXX_CDR_vYY.pdf	03/29/18
Demo Flight	Flight_XXXX.csv	06/09/18
PFR	Cansat2018_XXXX_PFR_vYY.pdf	06/10/18

XXXX is the team number. YY is the revision number. **Use this file format or your team will be removed from the competition. Files are to be in PDF format. No other formats will be accepted.**

At the end of the competition, the PDR, CDR, and PFR packages may be placed on the website for reference in subsequent years.

5.5 Slide Format Guidelines

The following guidelines shall be used when developing the presentation material:

- Use the template made available. Failure to do so will result in loss of points.
- All slides shall have simple white backgrounds. This helps reduce the file sizes and makes the slides easier to read.
- All slides shall have pages numbers in the footer. This is to allow for easier referencing of material during the reviews.
- All slides shall list the presenters name in the footer. This provides all the reviewers with an identity as to who is presenting the material.
- No embedded files or movies shall be included in the presentations. Not all reviewers will be able to access or view movies during the reviews due to network security settings at the various organizations involved.
- Each line-item in the review outlines shall correspond to a dedicated slide. This may result in slides with single bullets on them, however, this makes it easier for the reviewers to follow the presentation.

5.6 Disqualification Guidelines

The following are grounds for removal from the competition.

1. Any team found to be copying a PDR/CDR document from a previous competition will be disqualified from the competition.
2. Any team not meeting the basic competition requirements by CDR will be disqualified. This means teams using requirements from previous competitions and not following the current mission guide. This is an indication of teams copying a design from a previous competition.
3. Teams not demonstrating an understanding of the requirements of the competition will be disqualified.
4. Any team copying slides, content and designs from other teams will be disqualified.
5. Teams not submitting PDR and CDR documents in the required PDF format and file name convention as stated in the competition guide will be disqualified.
6. Teams not submitting PDR and CDR documents by the due dates will be disqualified.
7. Teams not scheduling presentation times by the specified date will be disqualified.
8. Teams not attending the PDR and CDR teleconference within 15 minutes of the set time will be disqualified. Find a phone that works. It can be a landline or mobile phone.
9. Excessive arguing with judges, mentors or any staff.

6. Flight Operations

6.1 Schedule

All times are referenced to central daylight time.

The competition starts Friday and ends Sunday evening.

Flight readiness review and safety inspection and preflight briefing will occur on Friday starting at 12 pm. The preflight briefing starts 7:30pm. Flight readiness reviews will continue up to 9 pm after the preflight briefing.

Saturday will be the launch day unless weather causes a postponement.

Sunday will be Post Flight Review presentations.

A detailed schedule will be provided at a later date. Be available Friday at noon until Sunday 8:00 pm.

6.2 Flight Readiness Review

Friday, teams are required to have their CanSats inspected for flight worthiness. Each team will be assigned a one half hour time slot to present their CanSat. This means the CanSat must be completed, built and ready to launch at the flight readiness review. A ball of wires and boards does not constitute flight ready. To emphasize, when presented, at the Flight Readiness Review (FRR), it must be in a state where it can be turned on and placed into a rocket for immediate launch and be fully operational.

6.2.1 FRR Sequence of Events

Teams must be prepared to demonstrate the ground station.

The first test will verify communications with the CanSat and demonstrate the ground station software. The ground station software operations will be scored at this time. The ground station must show data being plotted in real time.

The CanSat will be inspected for safety. The structure will be reviewed and determined if it is flight worthy. The mounting of the electronics and sensors will be reviewed. Mechanisms will be reviewed. Hazards will be identified such as heating elements exposed to the outside, etc.

The last test at the FRR will be the drop test. The CanSat must be in flight configuration and will be subjected to the drop test. If the test fails, the team must make repairs before being allowed to fly. The CanSat must pass the drop test in order to be launched.

If any CanSat is determined to not be flight ready, the team has until their flight the next day to make repairs and modifications. This is done to make sure your CanSat is completed before coming to the competition and for the safety of all people on the field.

Safety is highest priority. Any CanSat deemed not flight worthy will not be flown. The team

will lose all flight day points.

Crew assignments must be submitted at the flight readiness review in the Mission Operations Manual. The mission control officer will be given an identification so the flight coordinator and launch control officer knows who is the mission control officer.

The missions operations manual will be reviewed at the FRR.

6.3 Team Member Launch Operations Crew Assignments

Crew assignments must be submitted at the flight readiness review. The mission control officer will be given an identification so the flight coordinator and launch control officer knows who is the mission control officer.

The missions operations manual will be reviewed at the flight readiness review.
Team Member Launch Operations Crew Assignments

In order to have a successful launch, teams need to coordinate among themselves and with the flight coordinator. Team members need to be assigned to specific tasks and develop a checklist for a successful flight. The following task assignments must be delegated:

Mission Control Officer - This is a single person who is responsible for informing the Flight Coordinator when the team and their CanSat is ready to be launched.

Ground Station Crew - This is one or more persons who is responsible for monitoring the ground station for telemetry reception and issuing commands to the CanSat.

Recovery Crew - This is one or more persons responsible for tracking the CanSat and going out into the field for recovery and interacting with the field judges. This crew is responsible for making sure all field scores are filled in or loss of points will occur.

CanSat Crew - This is one or more persons responsible for preparing the CanSat, integrating it into the rocket, and verifying its status.

Team members can take on multiple roles except for the Mission Control Officer. The Mission Control Officer should be coordinating all efforts and interacting with the flight coordinator as needed. It is highly recommended that a checklist be developed that steps the crews through the preparation, integration, and flight operations.

Crew assignments must be submitted at the flight readiness review.

6.4 Mission Operations Manual

Each team is required to assemble a mission operations manual. The mission operations

manual includes five checklists/operations procedures to be created by the team. The checklists are for configuring the ground station, preparing the CanSat, and integrating the CanSat into the rocket. The launch preparation procedures, launch procedure, and removal procedure are provided. Additional steps can be added by the team. The document is available for download and modification. Each section of the mission operations manual must start on its own page. Pages should be numbered and a table of contents is to be included.

The team must have the mission operations manual assembled into a three ring binder. The mission control officer must use the manual during launch. The mission control operator shall go to the microphone at the launch site and announce their team and go through their launch procedure which will include the count down to the launch.

6.5 Launch Schedule

The launch will start at 1pm. All CanSats are to be submitted at noon. The time period of 8am to 12pm is available for launch preparations and check in. This is the time to set up antennas, ground stations, and final CanSat tests and preparations.

The launch will start at 1pm and will be done in groups of five. Each team will be assigned a round which will be scheduled in one half hour increments. Be prepared to load rockets toward the end of the previous round.

Each team will be given a thumb drive to upload their ground station data after they perform their flight operation. The thumb drive must be submitted to the judging table before leaving the field.

6.6 Competition Operations and Sequence of Events

Details of flight day operations shall be provided at the Pre-Flight Brief. An overview of the flight day operations include the following activities:

1. Arrive at launch site
2. Prepare CanSat for turn in. Make it flight ready and perform any tests.
3. Turn in CanSat at the check-in table by noon. It will be weighed and fit checked and stored in the off state until rocket preparation time.
4. Upon the team round, the team will collect their CanSat and load it into a rocket.
5. Verify the CanSat is communicating with the ground station.
6. Take the rocket with the ground station to the assigned launch pad. A staff member will install the rocket on the launch pad.
7. When it is time to launch, a judge will come by the ground station to monitor the ground station operation.
8. The team mission control officer will go to the launch control table and execute the launch procedures with the flight coordinator providing oversight.
9. Ground station crew will perform all required flight operations.
10. After all CanSats have launched for the current half hour round, team recovery personnel can head out to recover.
11. Ground station crew must clear out of the ground station area to allow the next round

ground stations to set up.

12. Ground station crew must turn in the thumb drive with any ground station data and or photos if desired to the check-in judge.
13. Recovery crew must return to the check-in for any final judging requirements.

Teams shall not touch the CanSat until the field judge verifies all necessary scoring information.

6.7 Second Flight Rules

Second flights are rare but do occasionally happen. The following conditions will qualify for a second flight.

1. If the rocket has a catastrophic failure. This includes the motor failing causing the rocket to crash or rocket parachute fails to deploy and crashes with the CanSat still inside. This event will require the team to have a working spare CanSat that can be prepared and made operational within an hour.
2. If the nose cone does not release keeping the CanSat from deploying and the CanSat has passed the fit check. Containers wider than 125mm and still flown will not get a second flight.

All are contingent on enough spare motors being available.

A team whose CanSat is destroyed due to a rocket failure will receive all launch day points only if they passed the drop test, the fit check, and the FRR tests and inspections otherwise no points will be received for the launch day.

6.8 Weather Delays

If weather conditions require the Saturday launch to be postponed to Sunday, the post flight reviews will be cancelled. Scores from PDR, CDR and the launch day will be used to determine the final rankings. Awards will be held Sunday evening.

If both Saturday and Sunday do not allow for launches, the final rankings will be based on PDR, CDR and flight readiness reviews.

Appendix A Field Safety Rules

1. Consumption of alcohol is not allowed.
2. Smoking is only allowed at designated areas. If anyone is caught smoking where it is not allowed, the landowner can throw you off the field.
3. Do not catch rockets or CanSats out of the air.
4. Stay behind the designated range line unless the range safety officer (RSO) or launch control officer (LCO) or flight coordinator has given permission to put your rocket on a pad.
5. Pay attention at all times. Every launch is potentially hazardous.
6. If a “heads up” launch is announced, you must be standing and facing the launch pad.
7. Do not retrieve a rocket from the range unless the LCO has given you permission.
8. Everyone must be alert when a “heads up!” is called and be ready to move.
9. Do not litter. Do not throw trash on the ground anywhere on the field. We have been invited to use the land owner's field and should treat it with respect. Any team caught throwing trash on the ground anywhere will be disqualified from the competition and the school will be notified of the disqualification. The landowner can order the team to leave the property and enforce the order.

Appendix B Presentation Recommendations

The following recommendations for presentation content and layout are being provided based on past experiences of the judges. These recommendations are not required to be followed but make it easier for the judges to review the material presented.

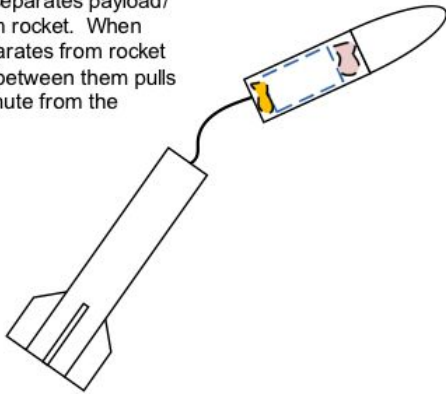
1. Use a consistent table format throughout the various subsystems when presenting requirements, component trades, and changes since previous reviews. Using a standard table format makes it easier for the judges to find the information in the table quickly since all tables are formatted the same.
2. During the CDR, the Changes Since PDR slides should use a table that contains a discussion of what the state of the design was at PDR, what it is at CDR, and what the rationale of the change was. Details of the change can be discussed in subsequent slides so an in-depth discussion is not always necessary.
3. Include the class year (freshman, sophomore, etc.) and major of each team member for reference. This doesn't play into the scoring of the team, however, it is often nice for the reviewer to know the status of the team members.
4. Be sure to follow the PDR and CDR outlines very carefully. Provide at least one chart for each scored item in the outline; this makes it easier for the judges to follow the presentation and confirm the required information is provided. In the presentation, be sure to address the questions and topics listed in the "description" column of the presentation outline -- those are the key points the judges are looking for.
5. Be clear which optional requirements, if any, are to be included in the design.
6. Be detailed in test descriptions. Identify specific tests, what is going to be done, and the pass/fail criteria.

Appendix C Payload Deployment Description

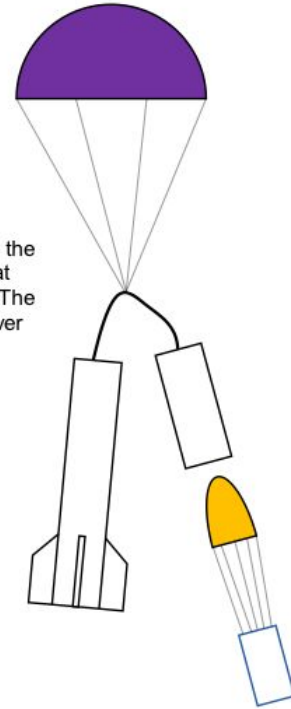
The figure illustrates a typical launch and separation sequence. Due to this nominal deployment sequence, it is recommended that CanSat be integrated with the payload section “upside down” such that the folded CanSat parachute rests on the payload section bulk plate. The CanSat then rests on the parachute and the folded nose cone parachute rests on the CanSat. Once the ejection charge burn is completed, the payload section and nose cone separate from the rocket and tip over. The nose cone slides out of the top of the payload section and the CanSat then falls out of the payload section due to gravity.

If a mission does not require a parachute at deployment from the rocket, the nadir pointing end of the cansat shall be facing toward the nose cone.

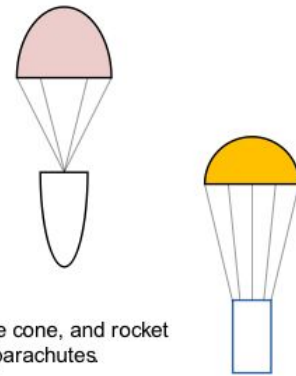
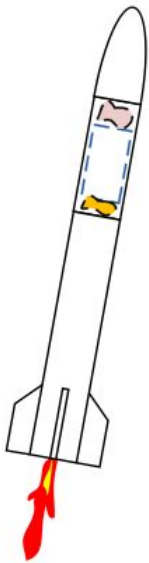
Ejection charge separates payload/nose section from rocket. When front section separates from rocket the shock chord between them pulls the rocket parachute from the rocket.



When the front section tips over, the nose cone falls off and the cansat falls out of the payload section. The cansat parachute now inflates over the cansat.



Cansat rests on its parachute. The nose cone parachute rests on the bottom of the cansat.



The cansat, nose cone, and rocket descend under parachutes.

Appendix D Acronyms

A	Analysis
CDR	Critical Design Review
CONOP	Concept of Operations
D	Demonstration
DCS	Descent Control System
FRR	Flight Readiness Review
GCS	Ground Control Station
HW	Hardware
HWR	Hardware Review
I	Inspection
LCO	Launch Control Officer
PDR	Preliminary Design Review
PFB	Pre Flight Briefing
PFR	Post Flight Review
RSO	Range Safety Officer
SOE	Sequence of Events
T	Test
TBD	To Be Determined
TBR	To Be Resolved
VM	Verification method

Appendix E Definitions

<i>Analysis</i>	Verification method that utilizes evaluation of data generated by accepted analytical techniques or simulations under defined conditions to show the item will meet the specified requirements.
<i>CDR</i>	A multi-disciplined technical review to ensure that the system under review can proceed into system fabrication, demonstration, and test; and can meet the stated performance requirements within cost (program budget), schedule (program schedule), risk, and other system constraints.
<i>CONOP</i>	Describes what the system will do and the way the system works from the operator's perspective. The CONOP is a high level description that should include a top-level block diagram.
<i>Demonstration</i>	Verification method that utilizes a qualitative exhibition of functional performance, usually accomplished with no or minimal instrumentation.
<i>Inspection</i>	Verification method that utilizes an examination of the item against applicable documentation to confirm compliance with requirements.
<i>Need Date</i>	Latest date a component or element (software, etc.) must be received or completed in order to not impact the end completion date.
<i>PDR</i>	A multi-disciplined technical review to ensure that the system under review can proceed into detailed design, and can meet the stated performance requirements within cost (program budget), schedule (program schedule), risk, and other system constraints.
<i>Shall</i>	Verb used to indicate a requirement is binding. All shall statements require verification.
<i>Should</i>	Verb used to define a goal or non-mandatory provision.
<i>Test</i>	Verification method utilizing operation of all or part of the item under controlled conditions, either real or simulated, to determine that the quantitative design or performance requirements have been met.
<i>To Be Determined</i>	An item or parameter that has not been specified at the time of document release.
<i>To Be Resolved</i>	An item or parameter that is preliminary or uncertain at the time of document release and for which a final value is to be specified at a later time.
<i>Validation</i>	Confirms that the system, as built (or as it will be built), satisfies the user's needs. Confirmation you built the right thing.
<i>Verification</i>	Confirms that the system, its elements, its interfaces, and incremental work products satisfy their requirements. Confirmation you built the

system right.

Will

Verb used to reference a binding or hard requirement elsewhere in the document text.

Appendix F – Payload Section

This information is provided to allow teams to build a test rocket payload section.

The payload section consists of a 20 inch long cardboard airframe tube, a coupler that slides into the air frame, a bulk plate and an eyebolt.

The top of the payload section is the open end where the cansat is inserted and the nose cone is inserted. The base of the payload section is the opposite end where the coupler is secured and the bulk plate is secured with the eyebolt.

Assembly sequence:

10. Epoxy the bulk plate to one end of the coupler tube.
11. Secure the eyebolt to the bulk plate. Place the washers on both sides of the bulk plate when securing the eyebolt.
12. Epoxy the coupler into the air frame with only half the coupler inserted in the airframe. Make sure no epoxy drips onto the exposed end of the coupler tube. It has to be clean and smooth for it to fit into the rocket.

Alternative to epoxying the coupler into the airframe is to use four screws or click-lock shank rivets.

Link to the payload kit:

<https://www.locprecision.com/product/payload-bays/>

Select the 5.38x20 inch payload

Link to the rocket kit:

<https://www.locprecision.com/product/minie-mag/>

